FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CIL HARDWARE

NUMBER: 02-2A-021114 -X

SUBSYSTEM NAME: FLIGHT CONTROL MECH - RUDDER SPEED BRAKE & BF

REVISION: 0 02/02/88

PART DATA

PART NAME

VENDOR NAME

PART NUMBER

VENDOR NUMBER

ASSY

: BODY FLAP ACTUATION

MC621-0056-0083

SRU : FILTER SCREEN

EXTENDED DESCRIPTION OF PART UNDER ANALYSIS:

FILTER SCREEN

REFERENCE DESIGNATORS:

QUANTITY OF LIKE ITEMS: 3

THREE (ONE EACH HYD SYSTEM)

FUNCTION:

SCREEN TYPE FILTERS ARE INCORPORATED AT THE VALVE MODULE PRESSURE PORTS P1, P2 AND P3 TO PREVENT INSOLUBLE CONTAMINANTS FROM ENTERING THE BODY FLAP SYSTEM.

FAILURE MODES EFFECTS ANALYSIS FMEA -- CIL FAILURE MODE

NUMBER: 02-2A-021114-02

REVISION#: 1

08/07/98

SUBSYSTEM NAME: FLIGHT CONTROL MECH - RUDDER SPEED BRAKE & BF

CRITICALITY OF THIS

LRU:

FAILURE MODE: 1R3

ITEM NAME: FILTER SCREEN

FAILURE MODE:

RESTRICTED OR NO FLOW

MISSION PHASE:

DO DE-ORBIT

VEHICLE/PAYLOAD/KIT EFFECTIVITY:

102 COLUMBIA

103 DISCOVERY 104 ATLANTIS

105 ENDEAVOUR

CAUSE:

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DEGRADED FILTER, CONTAMINATION

CRITICALITY 1/1 DURING INTACT ABORT ONLY? NO

REDUNDANCY SCREEN

A) PASS

B) FAIL

C) PASS

PASS/FAIL RATIONALE:

A)

B١

FAILS REDUNDANCY SCREEN "B" SINCE 100% RATE IS UNDETECTABLE BY CREW.

C)

- FAILURE EFFECTS -

(A) SUBSYSTEM:

LOSS OF ONE HYDRAULIC MOTOR RPM/TORQUE INPUT INTO SUMMER DIFFERENTIALS. (HYDRAULIC BRAKE IS APPLIED WITH ABSENCE OF HYDRAULIC FLUID PRESSURE, PREVENTING REMAINING TWO HYDRAULIC MOTORS FROM BACKDRIVING FAILED HYDRAULIC MOTOR). BODY FLAP OPERATES AT 100% RATE.

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(B) INTERFACING SUBSYSTEM(S):

NONE.

(C) MISSION:

NONE

(D) CREW, VEHICLE, AND ELEMENT(S):

LOSS OF MISSION, CREWIVEHICLE AFTER LOSS OF REMAINING TWO HYDRAULIC MOTORS, RESULTING IN LOSS OF BODY FLAP FUNCTION.

-DISPOSITION RATIONALE-

(A) DESIGN:

FILTER SIZED FOR VEHICLE LIFE. PROVEN DESIGN COMPATIBLE WITH REQUIREMENTS OF SE-F-0044 RATED 200 MICRON - ABSOLUTE - CLEANABLE - ELEMENT SIZED FOR WORST CASE CONTAMINANT LEVEL - 100 MISSION LIFE. UPSTREAM 5 MICRON MAIN FILTER AND SERVICING CART FILTER.

(B) TEST:

QUALIFICATION TESTS: PERFORMANCE, OPERATING LIFE, QUIESCENT LEAKAGE AND IMPULSE CYCLING. TESTED DURING POWER DRIVE UNIT (PDU) AND QUALIFICATION TESTING TO LEVEL 190.

ACCEPTANCE TESTS: PROOF PRESSURE, PROOF LOW PRESSURE, QUIESCENT LEAKAGE, FAILURE MODE TEST AND FUNCTIONAL TEST.

GROUND TURNAROUND TEST

ANY TURNAROUND CHECKOUT TESTING IS ACCOMPLISHED IN ACCORDANCE WITH OMRSD.

(C) INSPECTION:

RECEIVING INSPECTION

MATERIAL AND PROCESS CERTIFICATIONS VERIFIED. FABRICATION AND INSPECTION DATA ON FILE.

CONTAMINATION CONTROL

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FAILURE MODES EFFECTS ANALYSIS (FMEA) -- CIL FAILURE MODE NUMBER: 02-2A-021114- 02

FORMAL CONTAMINATION CONTROL PLAN VERIFIED. PROCESS AND CORROSION PROTECTION PROVISIONS VERIFIED. ASSEMBLY VERIFIED TO BE FREE OF PARTICULATES. FLUID SAMPLE REQUIREMENTS VERIFIED. CLEANING TO LEVEL 190 VERIFIED BY INSPECTION.

ASSEMBLY/INSTALLATION

ASSEMBLY AND INSTALLATION OPERATIONS VERIFIED BY SHOP TRAVELER MANDATORY INSPECTION POINTS (MIPS).

TESTING

ACCEPTANCE TESTS CERTIFICATION VERIFIED.

HANDLING/PACKAGING

HANDLING AND PACKAGING REQUIREMENTS ARE VERIFIED.

(D) FAILURE HISTORY:

CURRENT DATA ON TEST FAILURES, FLIGHT FAILURES, UNEXPLAINED ANOMALIES, AND OTHER FAILURES EXPERIENCED DURING GROUND PROCESSING ACTIVITY CAN BE FOUND IN THE PRACA DATA BASE.

(E) OPERATIONAL USE:

NONE

- APPROVALS -

EDITORIALLY APPROVED TECHNICAL APPROVAL

· BNA

: VIA APPROVAL FORM

J. Komusa 8-18-48

: 95-CIL-009 02-2A